

Advanced Modeling & Simulation (AMS) Seminar Series
NASA Ames Research Center, September 12th, 2019

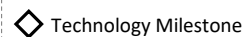
Turbulence Prediction in Aerospace CFD: Reality and the Vision 2030 Roadmap

Philippe Spalart and Mikhail Strelets



Background

- Discussion of the issues in physical modeling
 - Turbulence, not transition
- Emphasis on aircraft, and trend towards Certification by Analysis
 - Better, faster designs with less wind-tunnel time and no surprises
- Cruise condition under rather good control
 - Even buffet prediction is not impossible
- Hard regions of the envelope: high lift, stall, helicopters, landing gear...
- 2030 Roadmap was defined in 2014, with the following highlights:
 - Improved Reynolds-Stress models, 2018
 - Decision on continuing RANS research, 2019
 - Hybrid RANS-LES of high lift at flight Reynolds number, 2020
 - LES of high lift at flight Reynolds number, 2021
 - Demonstration on exascale machine, 2023
 - 30 exaflops by 2030



Technology Milestone



Technology Demonstration



Decision Gate

2015

2020

2025

2030

HPC

CFD on Massively Parallel Systems

PETASCALE

Demonstrate implementation of CFD algorithms for extreme parallelism in NASA CFD codes (e.g., FUN3D)

Demonstrate efficiently scaled CFD simulation capability on an exascale system

30 exaFLOPS, unsteady, maneuvering flight, full engine simulation (with combustion)

CFD on Revolutionary Systems (Quantum, Bio, etc.)

Demonstrate solution of a representative model problem

NO

NO

EXASCALE

Physical Modeling

RANS

Improved RST models in CFD codes

Highly accurate RST models for flow separation

Hybrid RANS/LES

Unsteady, complex geometry, separated flow at flight Reynolds number (e.g., high lift)

LES

Integrated transition prediction

WMLES/WRLES for complex 3D flows at appropriate Re

Combustion

Chemical kinetics calculation speedup

Chemical kinetics in LES

Unsteady, 3D geometry, separated flow (e.g., rotating turbomachinery with reactions)

Algorithms

Convergence/Robustness

Automated robust solvers

Grid convergence for a complete configuration

Multi-regime turbulence-chemistry interaction model

Production scalable entropy-stable solvers

Uncertainty Quantification (UQ)

Characterization of UQ in aerospace

Reliable error estimates in CFD codes

Uncertainty propagation capabilities in CFD

Large scale stochastic capabilities in CFD

Geometry and Grid Generation

Fixed Grid

Tighter CAD coupling

Large scale parallel mesh generation

Automated in-situ mesh with adaptive control

Adaptive Grid

Production AMR in CFD codes

Knowledge Extraction

Integrated Databases

Simplified data representation

Creation of real-time multi-fidelity database: 1000 unsteady CFD simulations plus test data with complete UQ of all data sources

Visualization

On demand analysis/visualization of a 10B point unsteady CFD simulation

On demand analysis/visualization of a 100B point unsteady CFD simulation

MDAO

Define standard for coupling to other disciplines

High fidelity coupling techniques/frameworks

Robust CFD for complex MDAs

Incorporation of UQ for MDAO

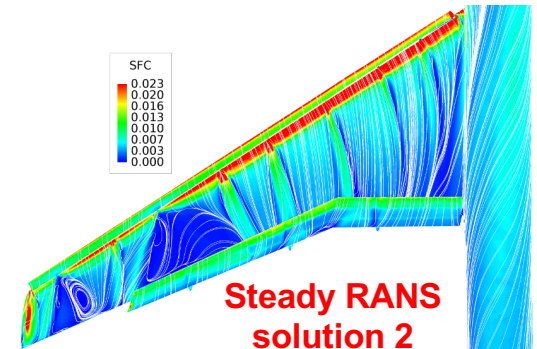
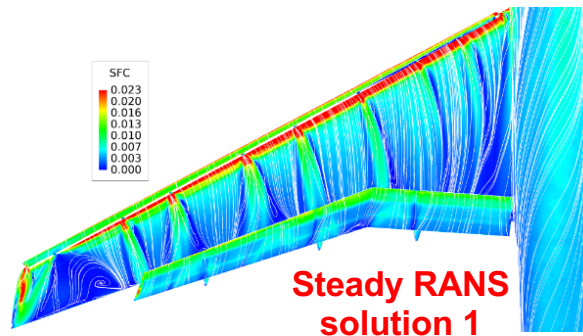
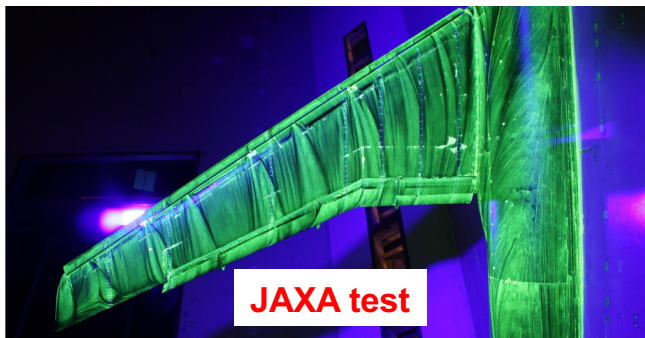
MDAO simulation of an entire aircraft (e.g., aero-acoustics)

UQ-Enabled MDAO

Achievements, 2014 to 2019

- Reynolds-Stress Models established in DLR and NASA codes.
 - Highly accurate? No. RSM's do not consistently improve over eddy-viscosity models
 - Especially SARC-QCR (-:
 - Convergence can be difficult
 - Models are almost static
 - Brief efforts by Rumsey and Spalart to alter SSG part not fruitful
 - See Eisfeld papers at this meeting
 - This appears to settle the “2019 Decision Gate”
- The Turbulence Modeling Benchmark Discussion Group, in a white paper, objects to stopping RANS research (AIAA-2019-0317, Bush et al.)
 - Cost of turbulence-resolving methods
 - Wide expectations that Artificial Intelligence will revolutionize RANS field
- Steady RANS models and codes plagued by multiple solutions
 - Worst symptom is “pizza slice” wide separation behind slat brackets
 - Insensitive to model and algorithm, much more sudden than in wind tunnel

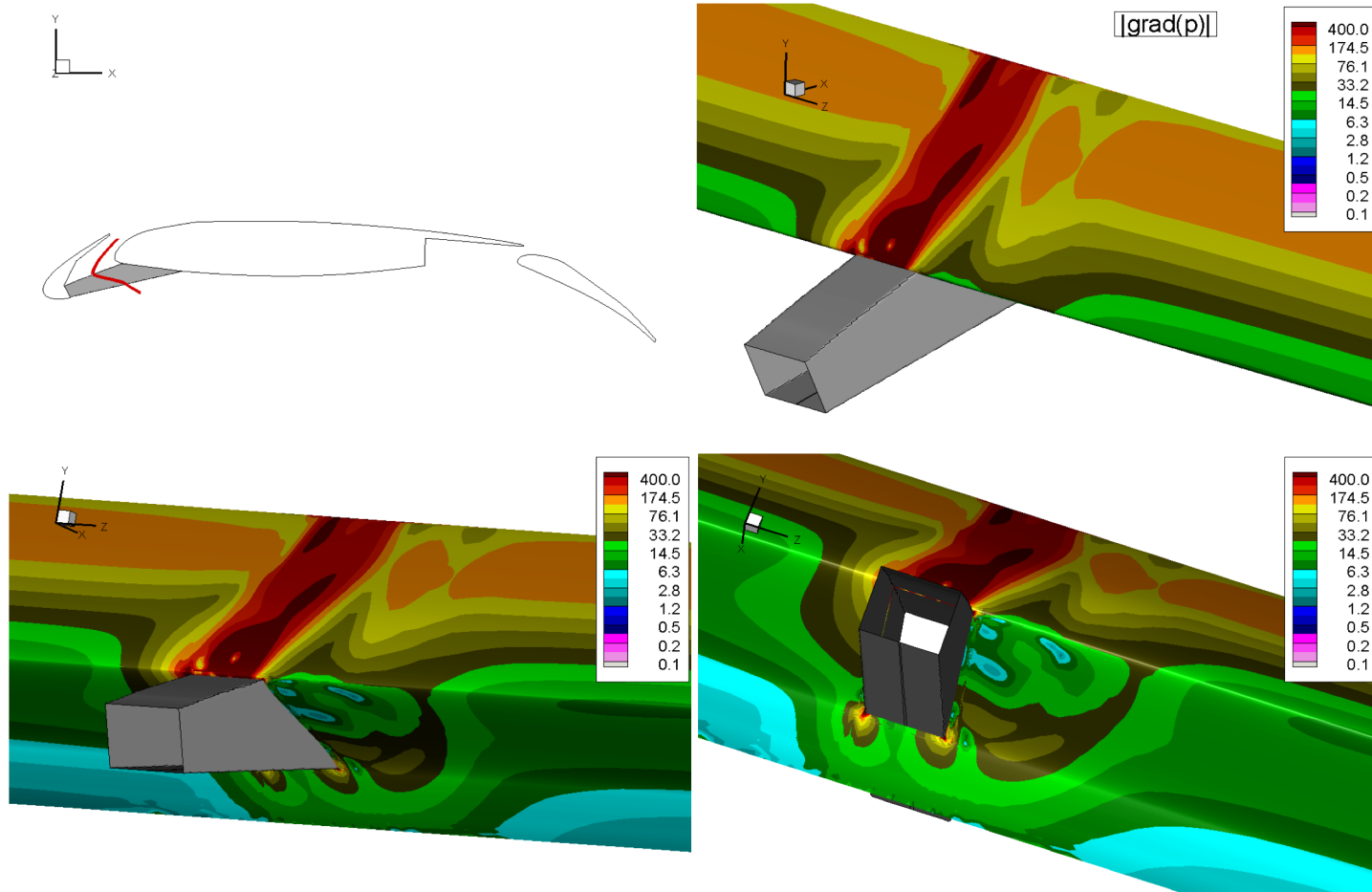
The Slat-Bracket Problem (“Pizza Slice”)



AIAA 2018-1037. Cary, Mani, Yousuf, & Li

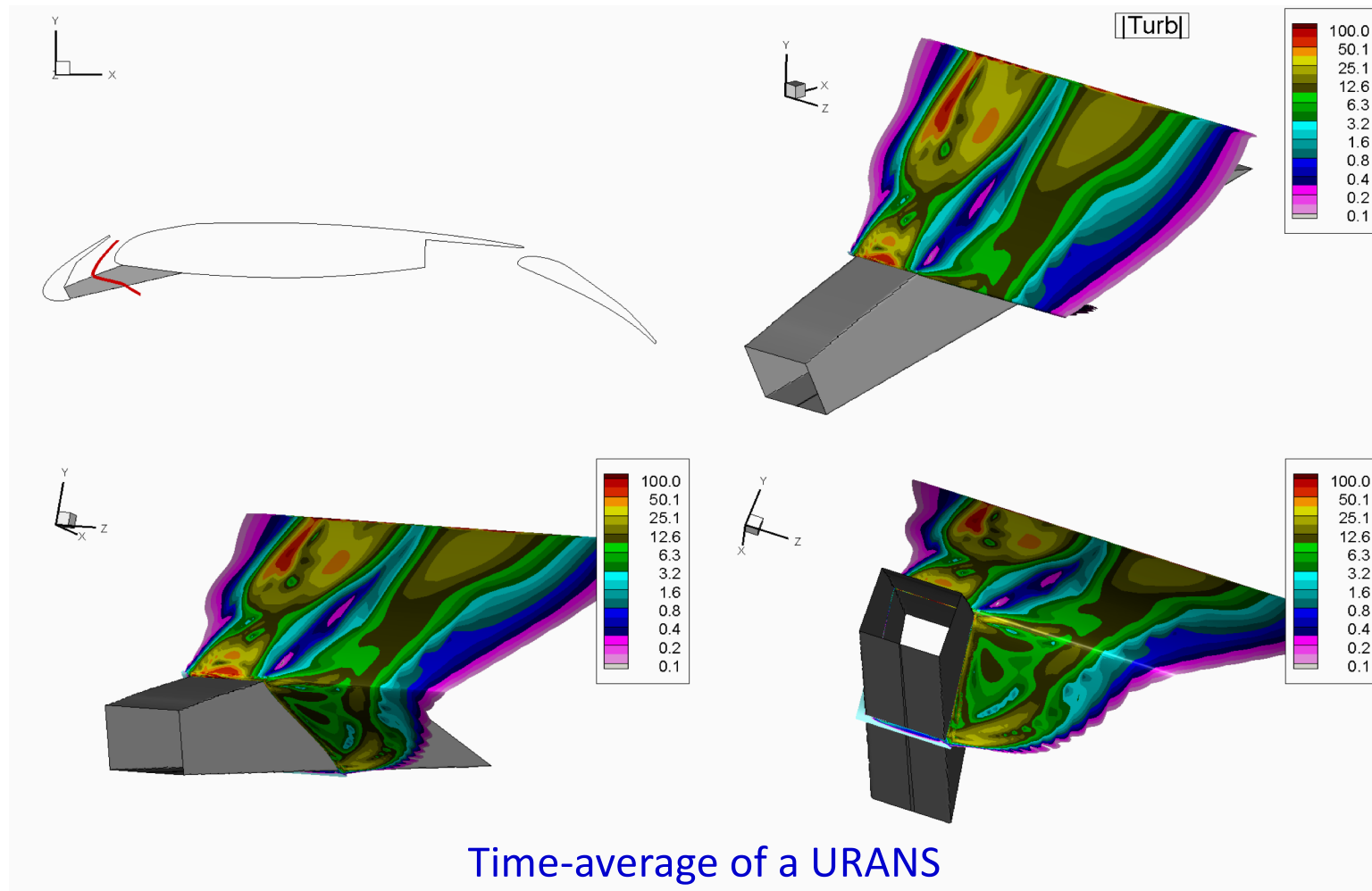
- The key question: can RANS models be made to work well enough?
- The phenomenon appears to be spurious, or at least premature
- It cuts across turbulence models and codes
- It is agreed that we don't have grid convergence, but grid adaptation failed to suppress it
- Is it a “robust consequence” of the steady RANS equations?
- Do the models **cause** it, or are they only too weak to suppress it?
- Is the bracket region “violently 3D, essentially convecting and rotating vorticity?”

Pressure-Gradient Term in Momentum Equation



Time-average of a URANS

Reynolds-Stress Term in Momentum Equation



Turbulence Models in Simple Flow

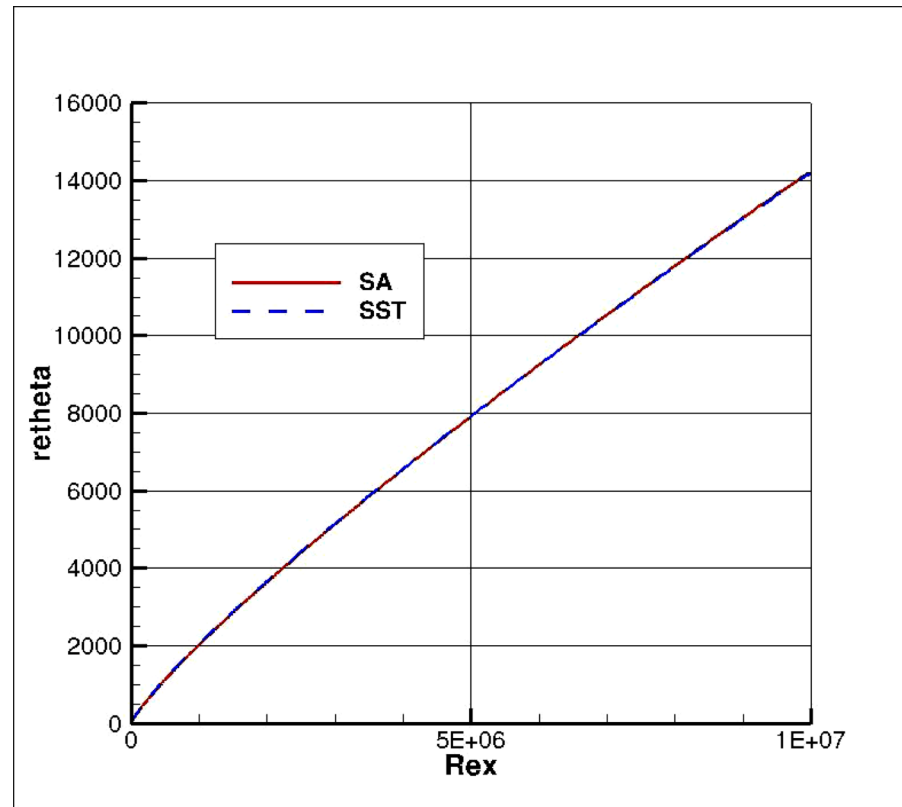


Figure from TMR



◇ Technology Milestone

★ Technology Demonstration

⊕ Decision Gate

2015

2020

2025

2030

HPC

CFD on Massively Parallel Systems

PETASCALE

CFD on Revolutionary Systems
(Quantum, Bio, etc.)

Demonstrate implementation of CFD algorithms for extreme parallelism in NASA CFD codes (e.g., FUN3D)

Demonstrate efficiently scaled CFD simulation capability on an exascale system

30 exaFLOPS, unsteady, maneuvering flight, full engine simulation (with combustion)

Demonstrate solution of a representative model problem

NO

NO

YES

EXASCALE

Physical Modeling

RANS

Improved RST models in CFD codes

Highly accurate RST models for flow separation

Hybrid RANS/LES

Integrated transition prediction

Unsteady, complex geometry, separated flow at flight Reynolds number (e.g., high lift)

LES

WMLES/WRLES for complex 3D flows at appropriate Re

Combustion

Chemical kinetics calculation speedup

Chemical kinetics in LES

Unsteady, 3D geometry, separated flow (e.g., rotating turbomachinery with reactions)

Algorithms

Convergence/Robustness

Automated robust solvers

Grid convergence for a complete configuration

Multi-regime turbulence-chemistry interaction model

Production scalable entropy-stable solvers

Uncertainty Quantification (UQ)

Characterization of UQ in aerospace

Reliable error estimates in CFD codes

Uncertainty propagation capabilities in CFD

Large scale stochastic capabilities in CFD

Geometry and Grid Generation

Fixed Grid

Tighter CAD coupling

Large scale parallel mesh generation

Automated in-situ mesh with adaptive control

Adaptive Grid

Production AMR in CFD codes

Knowledge Extraction

Integrated Databases

Simplified data representation

Creation of real-time multi-fidelity database: 1000 unsteady CFD simulations plus test data with complete UQ of all data sources

Visualization

On demand analysis/visualization of a 10B point unsteady CFD simulation

On demand analysis/visualization of a 100B point unsteady CFD simulation

MDAO

Define standard for coupling to other disciplines

High fidelity coupling techniques/frameworks

Incorporation of UQ for MDAO

Robust CFD for complex MDAs

MDAO simulation of an entire aircraft (e.g., aero-acoustics)

UQ-Enabled MDAO



◇ Technology Milestone

★ Technology Demonstration

⊕ Decision Gate

2015

2020

2025

2030

HPC

CFD on Massively Parallel Systems

PETASCALE

CFD on Revolutionary Systems
(Quantum, Bio, etc.)

Demonstrate implementation of CFD algorithms for extreme parallelism in NASA CFD codes (e.g., FUN3D)

Demonstrate efficiently scaled CFD simulation capability on an exascale system

30 exaFLOPS, unsteady, maneuvering flight, full engine simulation (with combustion)

Demonstrate solution of a representative model problem

NO

NO

YES

EXASCALE

Physical Modeling

RANS Improved RST models in CFD codes

Highly accurate RST models for flow separation

Hybrid RANS/LES

Unsteady, complex geometry, separated flow at flight Reynolds number (e.g., high lift)

LES

Integrated transition prediction

VMLES/WRLES for complex 3D flows at appropriate Re

Combustion

Chemical kinetics calculation speedup

Chemical kinetics in LES

Unsteady, 3D geometry, separated flow (e.g., rotating turbomachinery with reactions)

Algorithms

Convergence/Robustness

Automated robust solvers

Grid convergence for a complete configuration

Multi-regime turbulence-chemistry interaction model

Production scalable entropy-stable solvers

Uncertainty Quantification (UQ)

Scalable optimal solvers

Large scale stochastic capabilities in CFD

Geometry and Grid Generation

Fixed Grid

Tighter CAD coupling

Reliable error estimates in CFD codes

Uncertainty propagation capabilities in CFD

Automated in-situ mesh with adaptive control

Adaptive Grid

Production AMR in CFD codes

Knowledge Extraction

Integrated Databases

Simplified data representation

Creation of real-time multi-fidelity database: 1000 unsteady CFD simulations plus test data with complete UQ of all data sources

Visualization

On demand analysis/visualization of a 10B point unsteady CFD simulation

On demand analysis/visualization of a 100B point unsteady CFD simulation

MDAO

Define standard for coupling to other disciplines

High fidelity coupling techniques/frameworks

Incorporation of UQ for MDAO

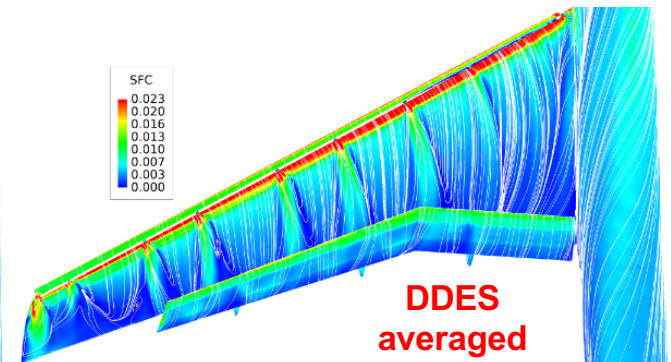
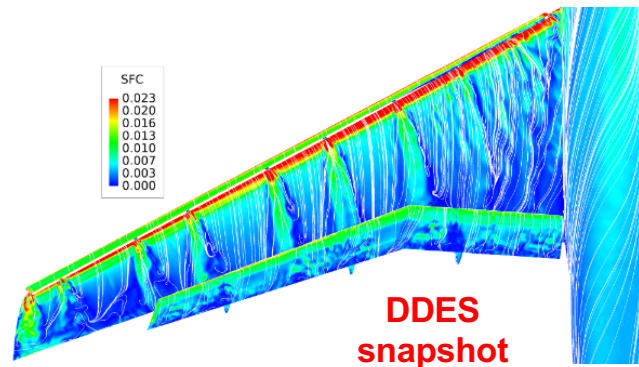
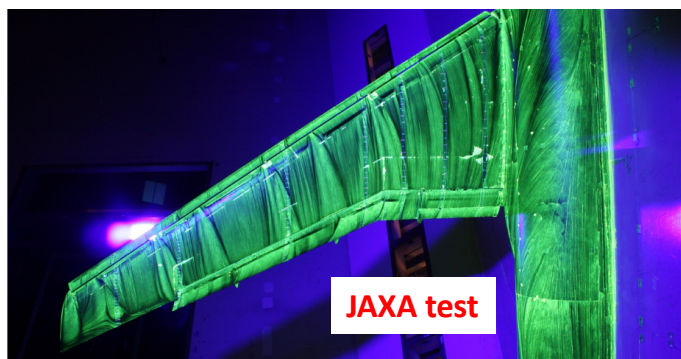
Robust CFD for complex MDAs

MDAO simulation of an entire aircraft (e.g., aero-acoustics)

UQ-Enabled MDAO

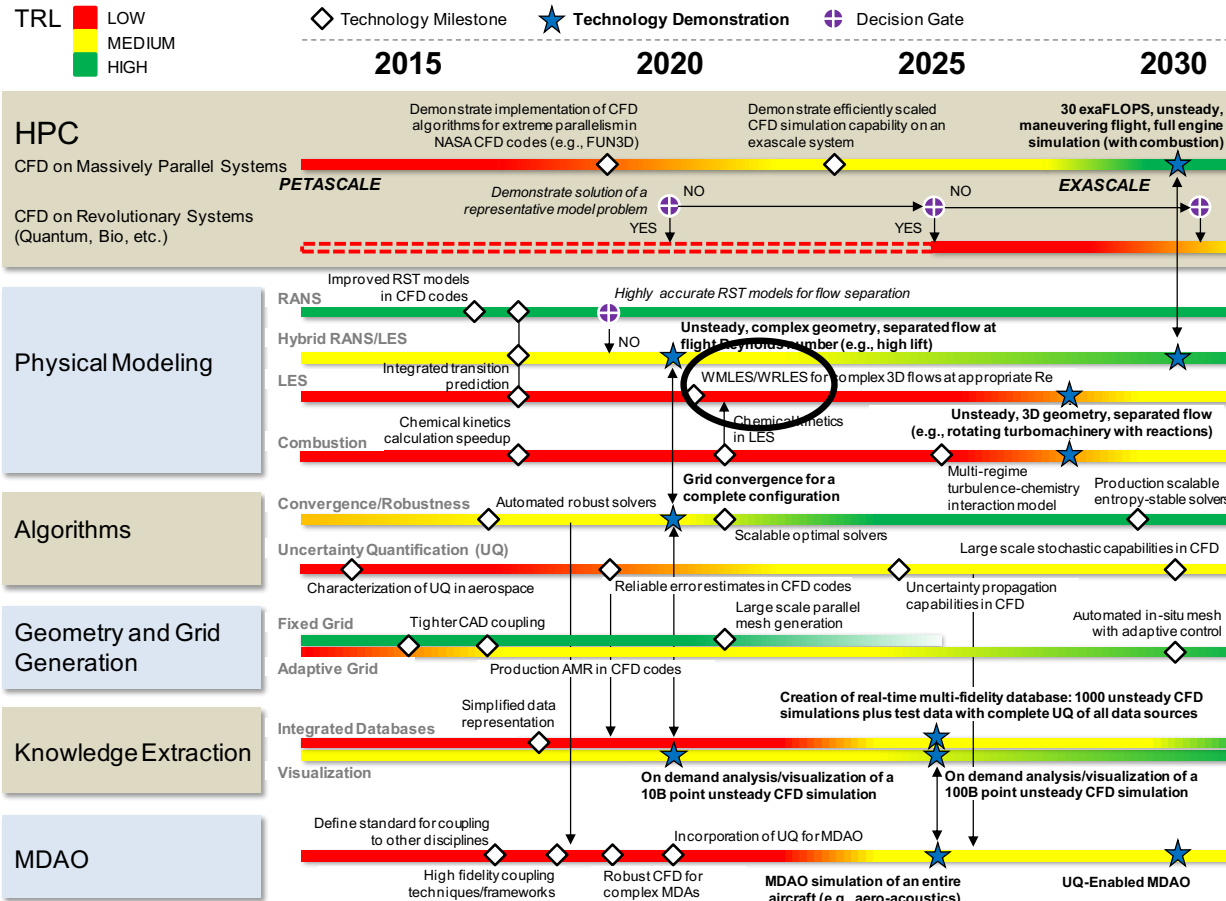
Preliminary Success of DDES at 3rd High-Lift Workshop

- Turbulence-resolving approaches appear immune to pizza slice issue
 - DES, WMLES, LBM-VLES...
- They tend to give better lift than RANS near C_{lmax}
- No reports of multiple solutions, from “cold starts”



AIAA 2018-1037. Cary, Mani, Yousuf, & Li

CFD Vision 2030 Roadmap

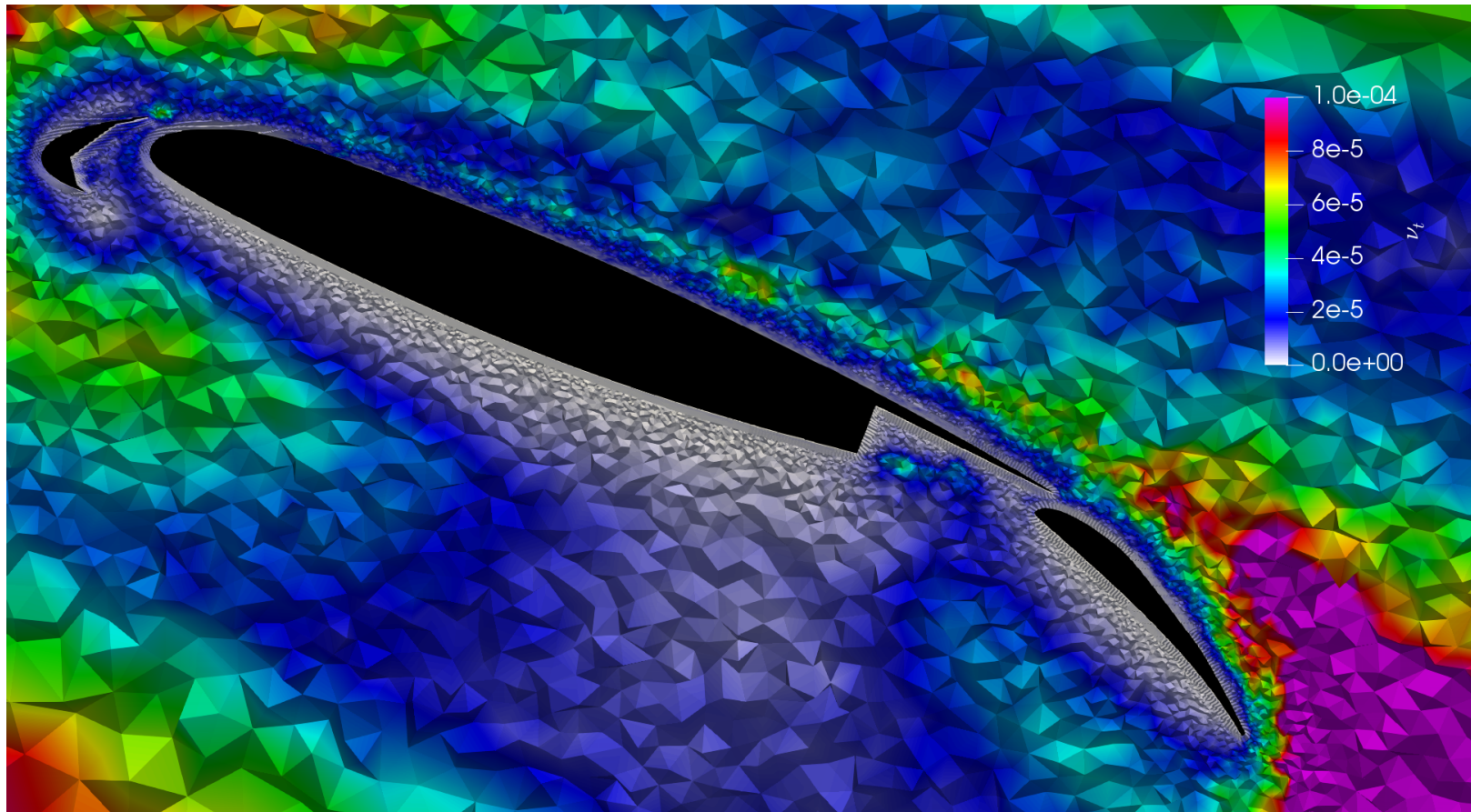


2014-2019 Activity: Wall-Modeled Large-Eddy Simulation

- Some people consider WMLES to be “turn key,” just expensive
 - The “N_{cubes} Problem” remains: in the thinner BL regions, the WM does everything
- It has given encouraging results for high lift, compared with RANS
 - Particularly at Stanford and Barcelona; PowerFLOW and PHASTA are similar
- For simple shear flows, channel and TBL, WM is the key difficulty
 - The SGS model proper has been validated, and is not very sensitive
- *Not so* for external flows with “real” geometries
$$\nu_{SGS} = f(S_{ij}, \text{grid cell})$$
 - This is innocuous at the end of the inertial range in a turbulent region
- Real flows have strain and grid variations in regions that should be inviscid and irrotational
 - The non-uniform SGS viscosity then creates vorticity
 - However, numerical errors also do... (private comments of Lehmkuhl and Rodriguez)
- The SA and SST-V models, used in DES, do not have this problem
 - The PDE also improves the smoothness of the eddy viscosity
 - ILSA also seems largely immune

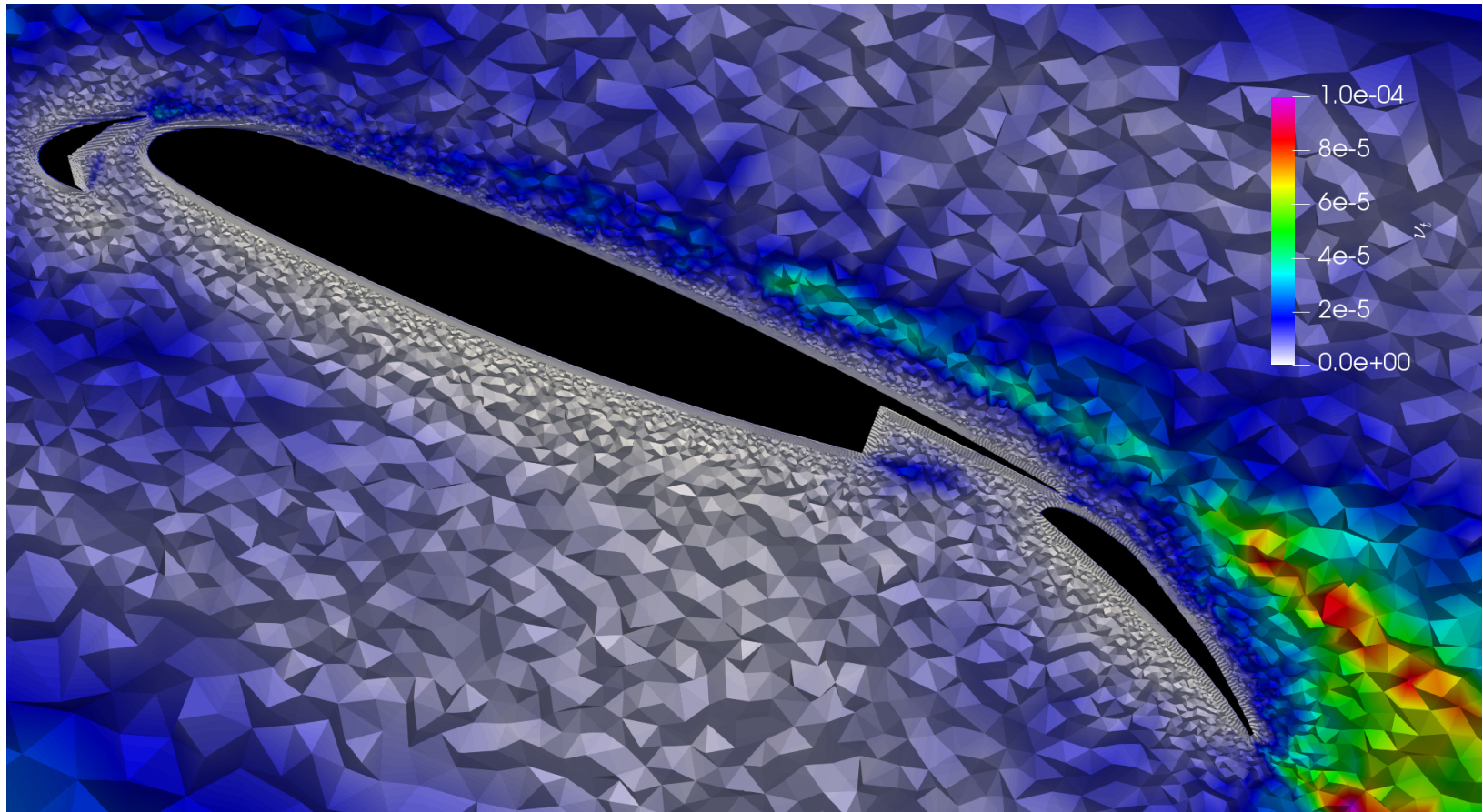
SGS Eddy Viscosity of Vreman Model

- Work of O. Lehmkuhl, Alya code



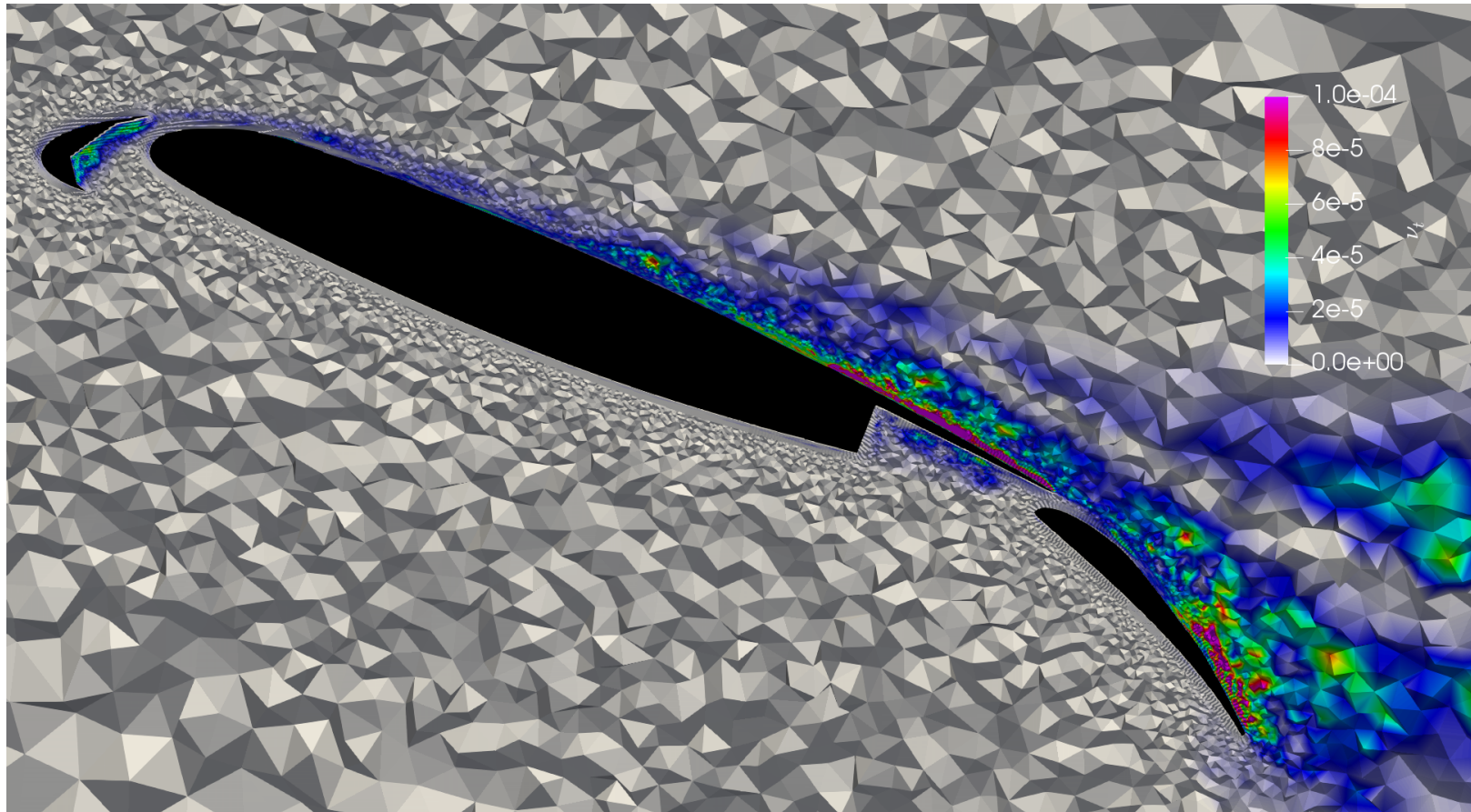
SGS Eddy Viscosity of Smagorinsky Model

- Work of O. Lehmkuhl

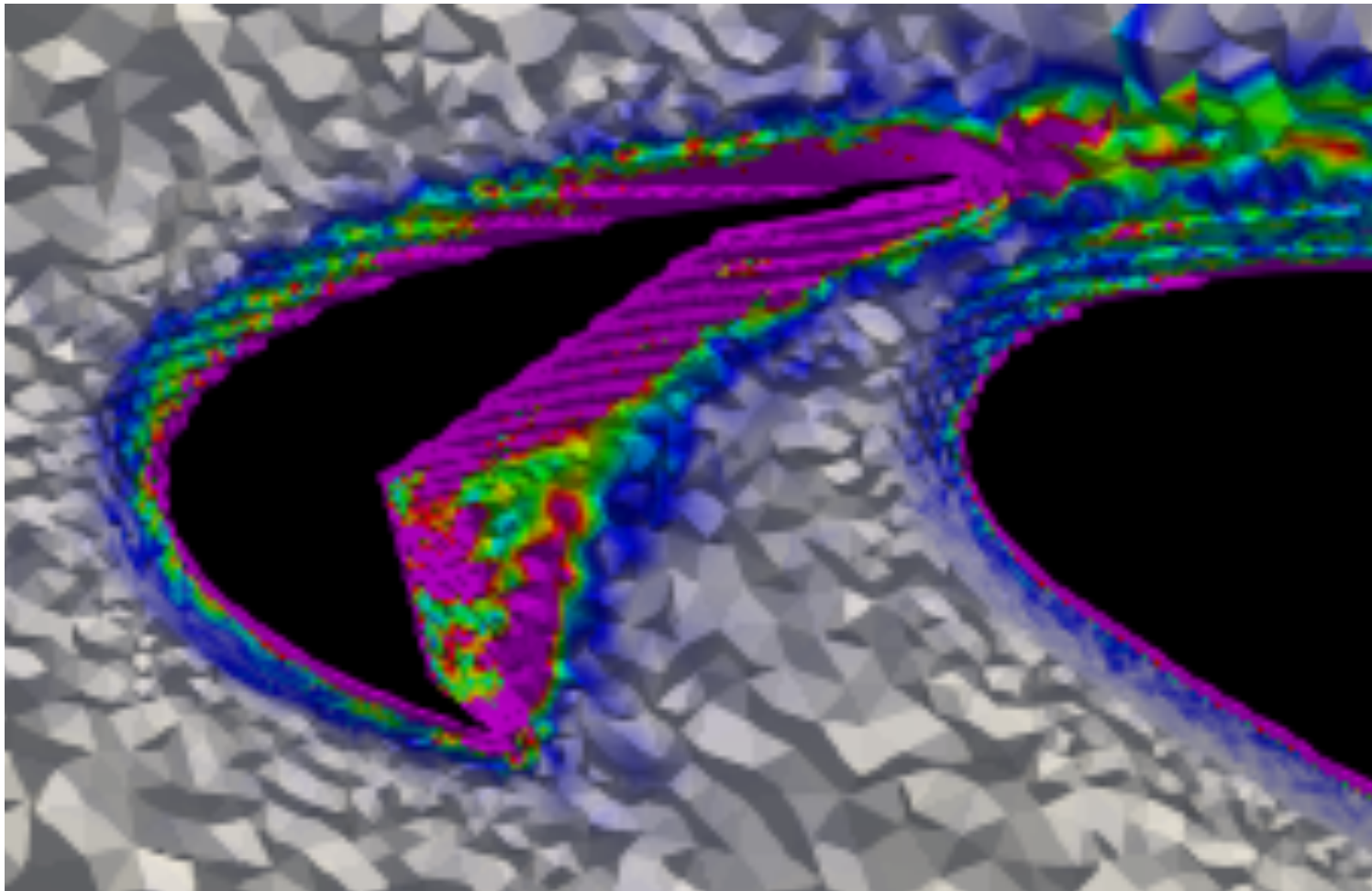


SGS Eddy Viscosity of ILSA Model

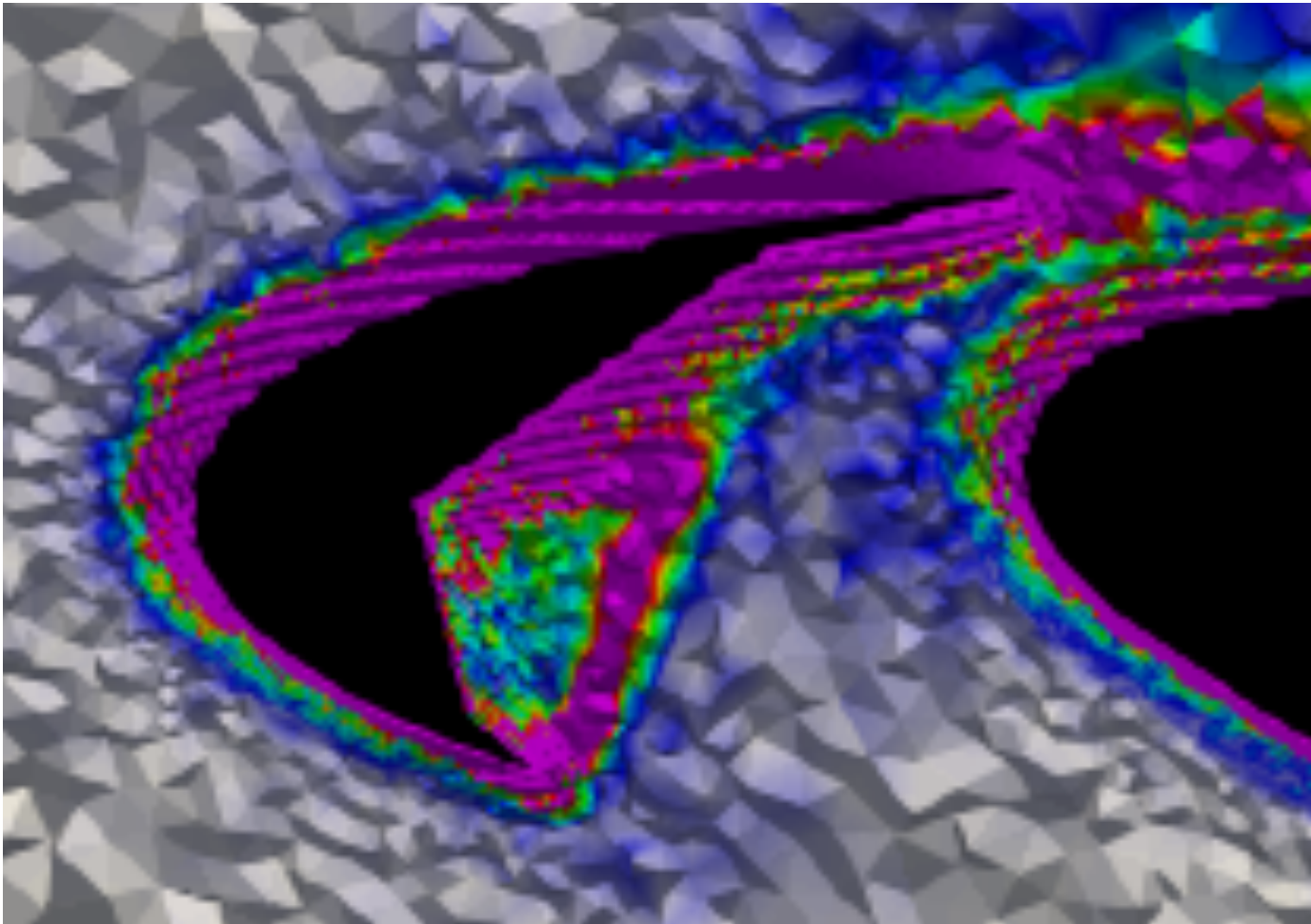
- Work of O. Lehmkuhl with U. Piomelli



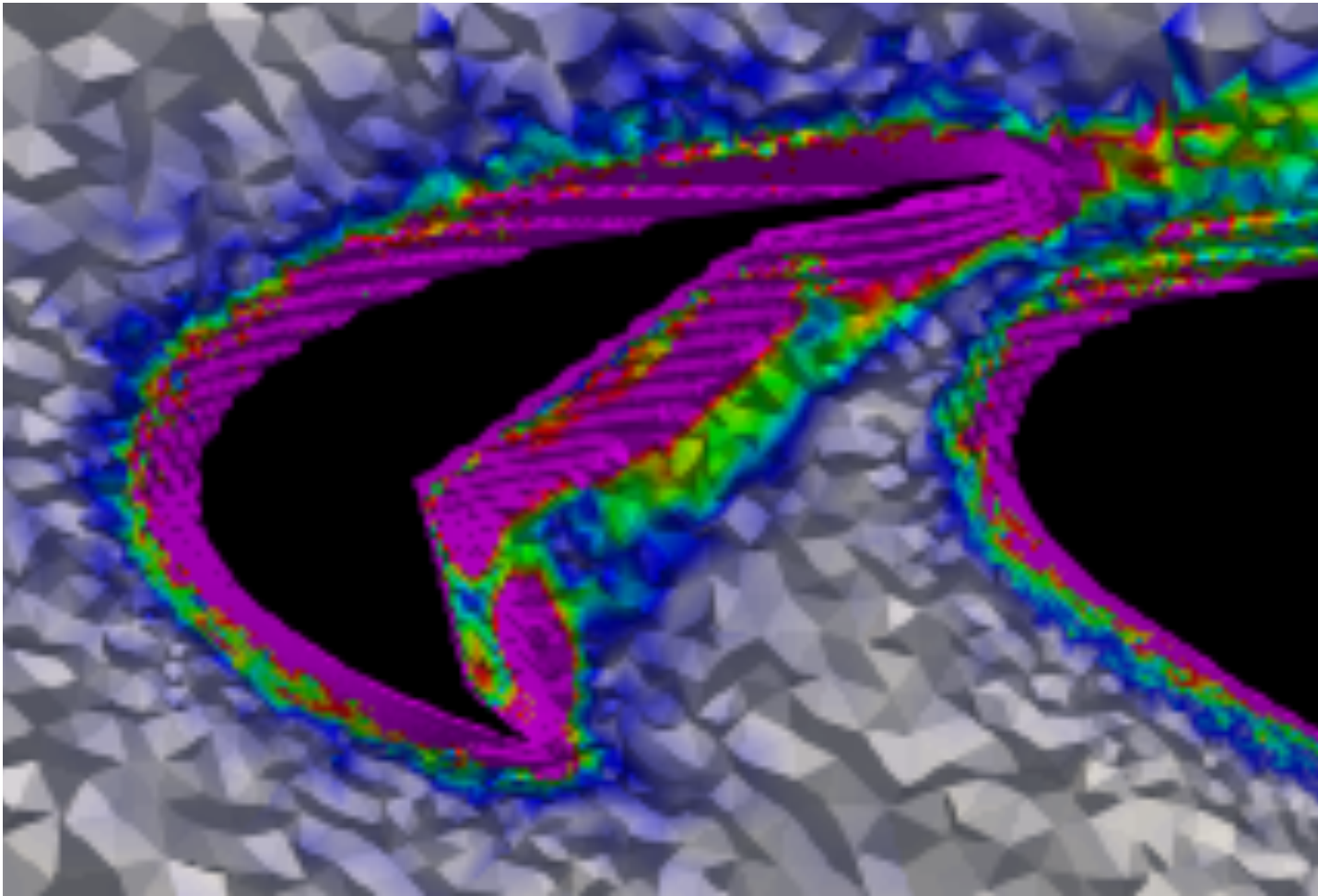
Vorticity with Vreman Model

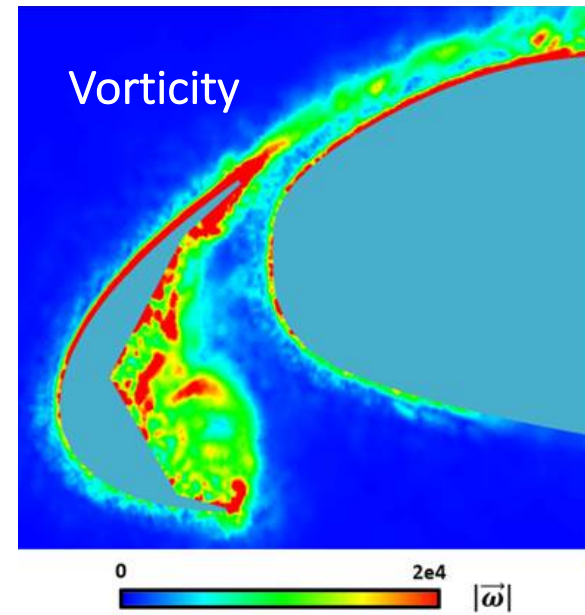
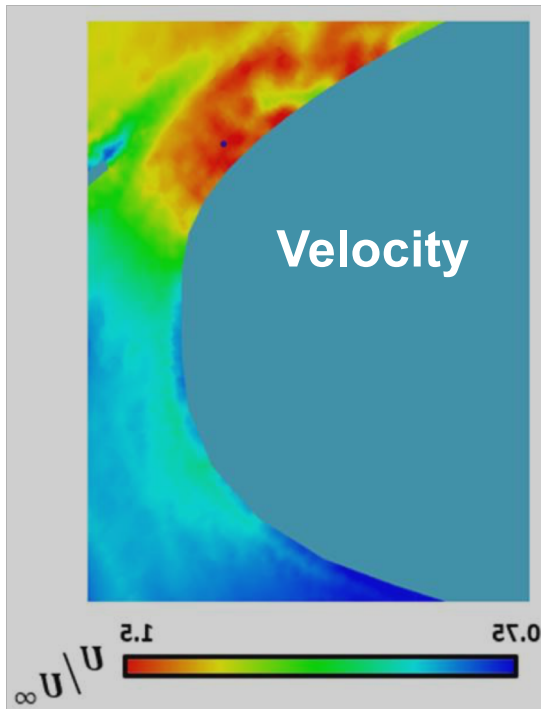


Vorticity with Smagorinsky Model

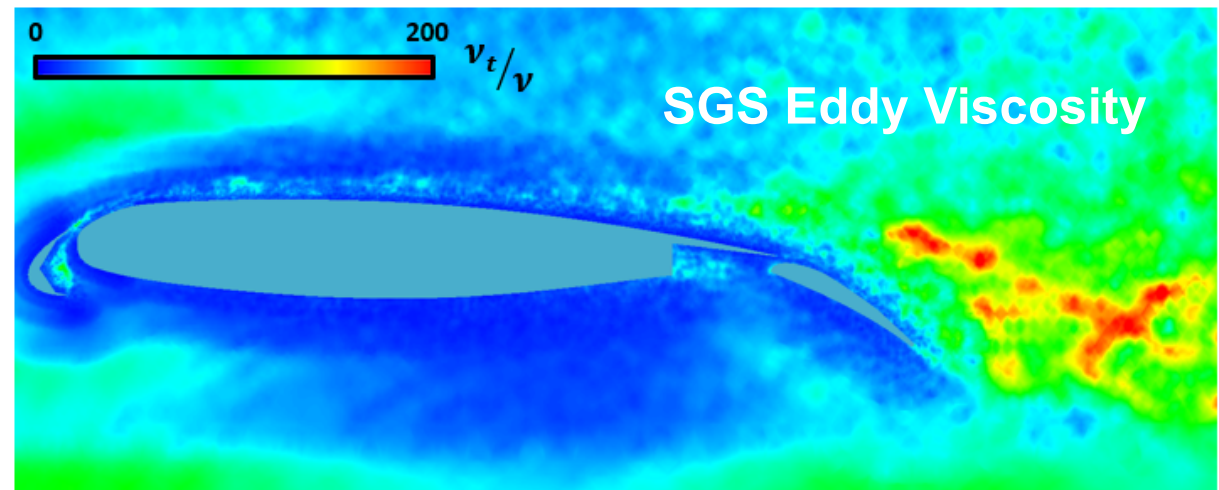


Vorticity with ILSA Model



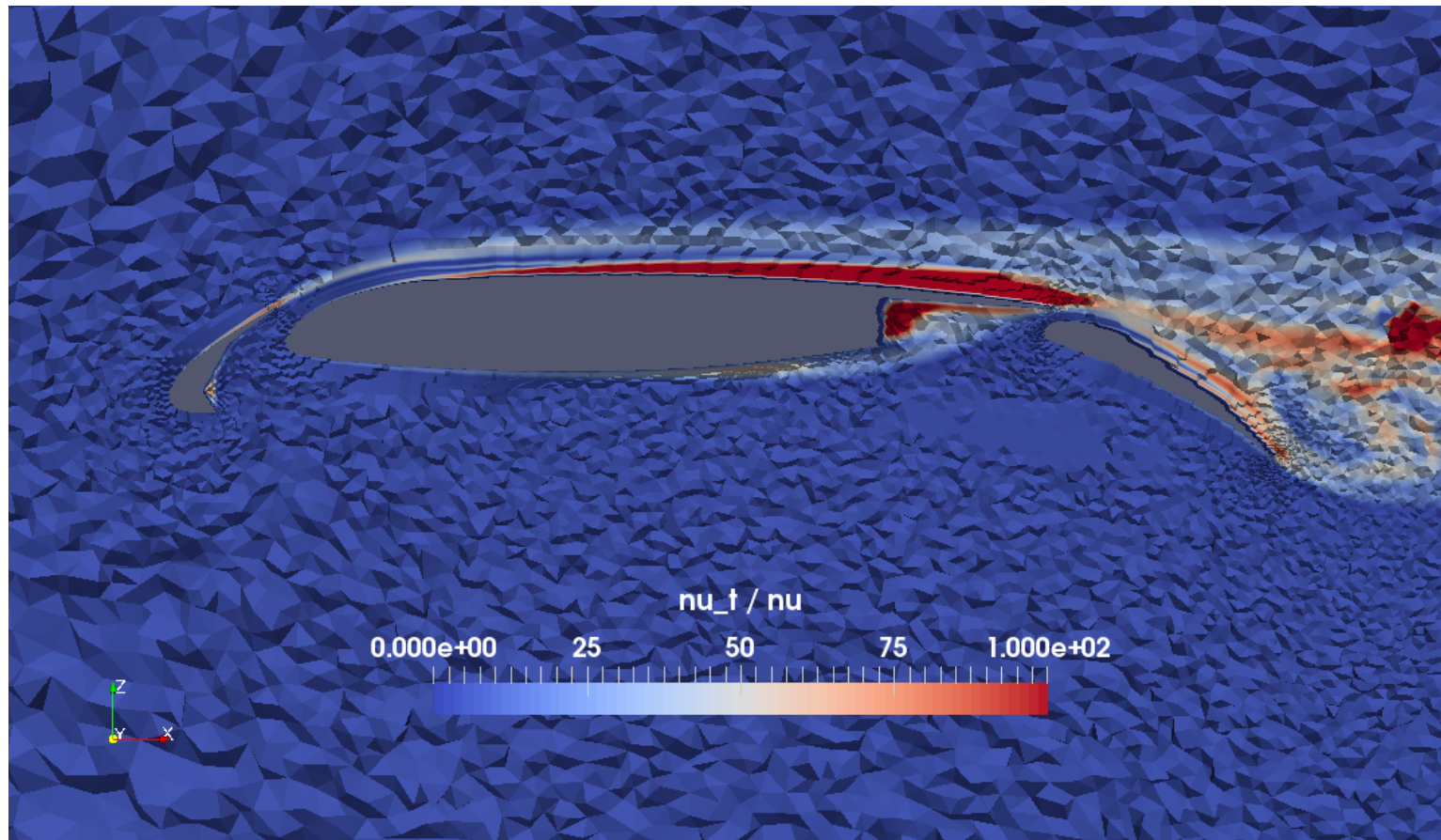


- Work of K. Goc and P. Moin
 - CharLES code, Vreman model
 - Slip WM



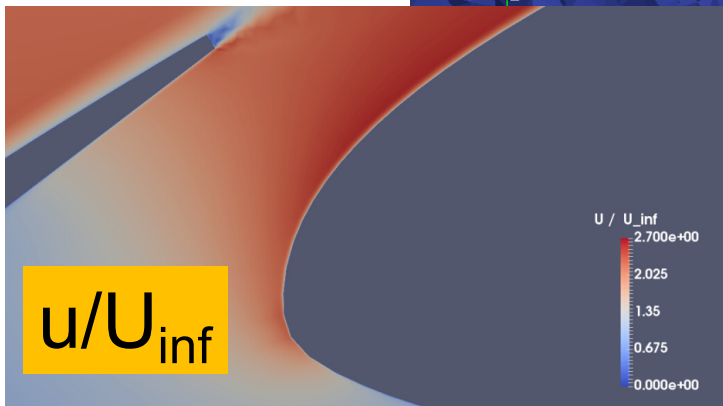
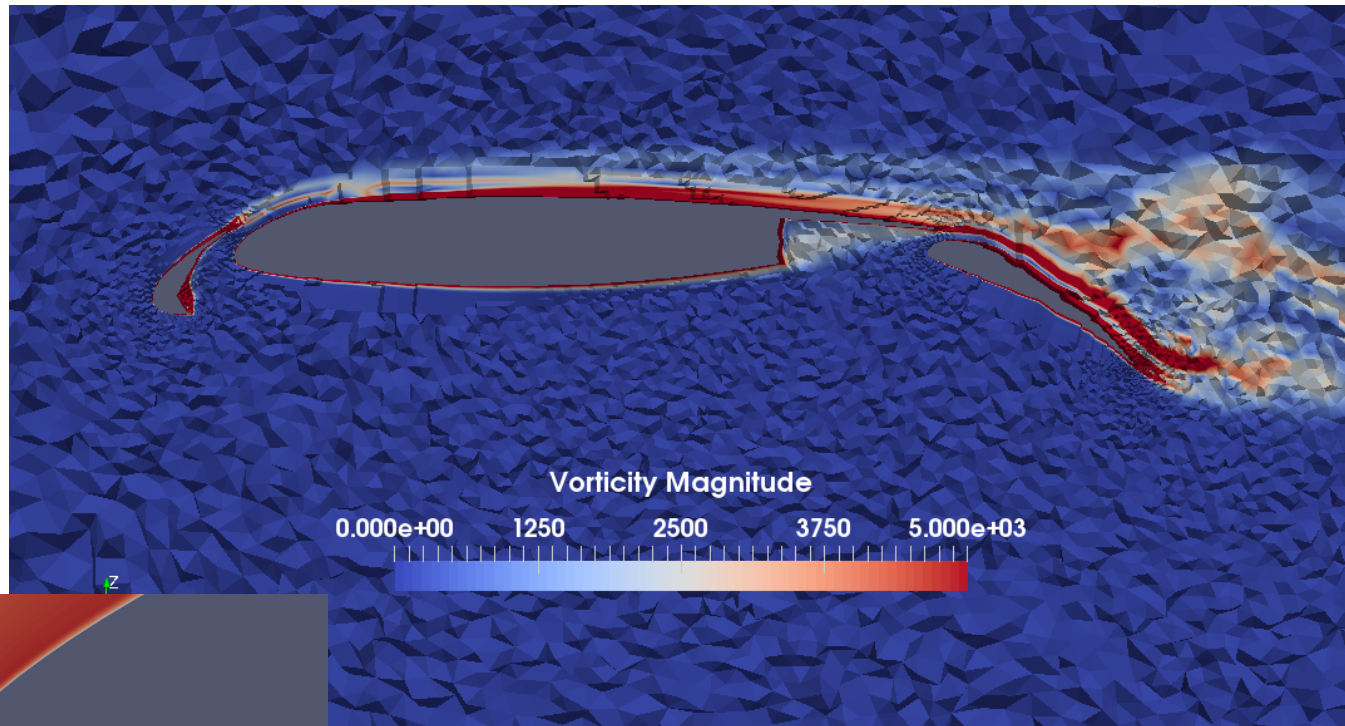
DDES Eddy Viscosity

- Work of R. Balin and K. Jansen, PHASTA code



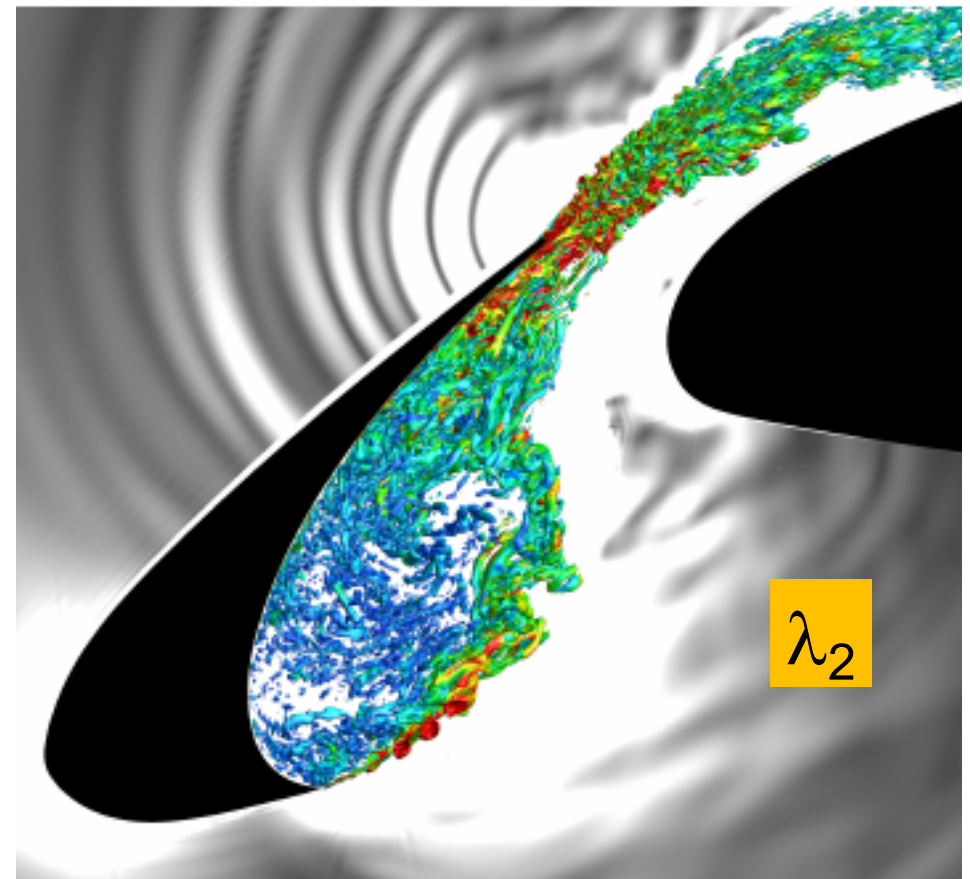
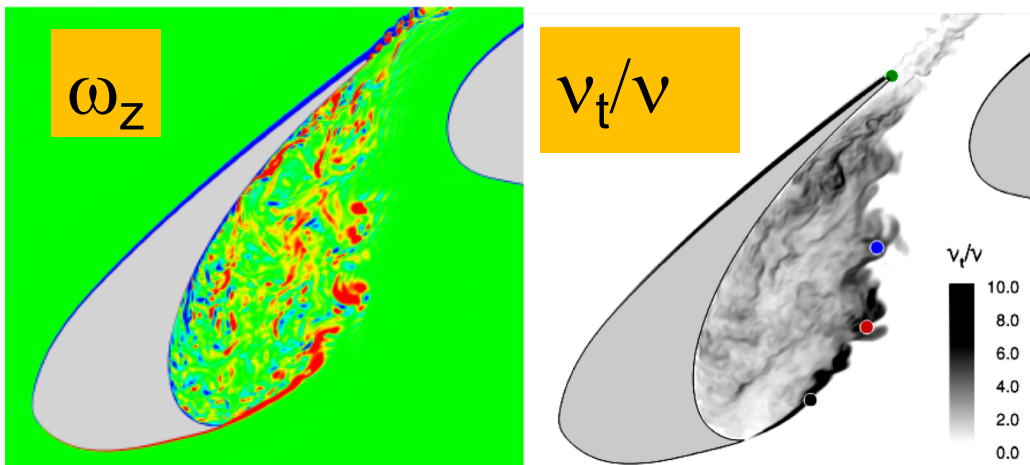
DDES Vorticity

- Balin,
Jansen,
PHASTA



Narrow Slice Simulation

- Work of T. Knacke and F. Thiele (2013-2162)
- DDES in ELAN code
- Width 3.3% of chord (30P30N)
 - 9M points in cove
 - Time sample 70,000 steps, $T \sim 8 c / U$
- Real problem is hundreds of times larger
 - Lemkuhl had 70M points and 193M elements for the half airplane



Artificial Intelligence in Turbulence Modeling

- AI has made great strides in extremely difficult areas such as translation
 - Tools proposed here include Machine Learning, Big Data, Deep Neural Networks, etc.
 - Many paper titles sound like: “Physics-Informed Machine Learning Approach for Augmenting Turbulence Models: A Comprehensive Framework”
- RANS modeling arguably has stagnated for decades
 - In Aerodynamics. Not as much in internal flows?
 - It’s possible that RANS modeling faces a “Fundamental Paradox” and has an “Accuracy Barrier,” and the community’s expectations/the demands of CFD are not realistic (local model formulation)
 - The SA and SST models are very useful, but not founded on theory or DNS
- There is logic in hoping AI can end the stagnation, with two threads:
 - 1. New thinking, new terms, new physics, some based on DNS data
 - 2. More powerful optimization of existing models over a wide range of flows
- Should this include “historical” modelers, or start from a “clean sheet of paper?”
 - Many “clean sheet” efforts violate Galilean Invariance, or have more subtle defects
 - A very clear “mission” must exist
 - Very few code-ready new models, or model versions, have been produced so far
 - Except by Weatheritt & Sandberg, using Genetics of the equations!
 - Note that Symbolic Manipulation of equations has not caused much progress
- A large European proposal, HiFi-TURB of Hirsch & Haase, hinges on this hope
 - Kick-off meeting in July 2019! Historical modelers very much included, and NASA

Summary

- Since 2014, our community's work has been collaborative and smart enough
 - Experimentalists, numerical types, and modelers
- Budgets are not matching the value of and the promises made for CFD
- The growth of computing power has slowed badly
- For high-lift, modeling can still hide behind the lack of grid convergence
 - Yet, it is certain modeling will become the "tent pole," in the steady RANS setting
- Traditional turbulence modeling is challenged from two sides:
 - Turbulence-resolving simulations
 - These are promising, but far from industry practical. We need **many** exaflops
 - The flow fields have some very "interesting" features...
 - We contend that DES is cleaner, and will deliver well before WMLES and VLES
 - Artificial intelligence
 - We contend that this work is still in its infancy, and much of it is simply unsuccessful
 - A lot of "adult supervision" is needed
 - Did we the "adults" fail to explain modeling (too bad Wilcox's book is now rare)?
- Several of the Vision 2030 milestones will be missed